

Sound and Automated Synthesis of Digital Controllers for Continuous Plants

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March 2017

Cyber-Physical Systems (CPS)

- ▶ modern controls are implemented with digital microcontrollers, embedded within dynamical plants representing physical components
- ▶ digital control literature: success and limitations



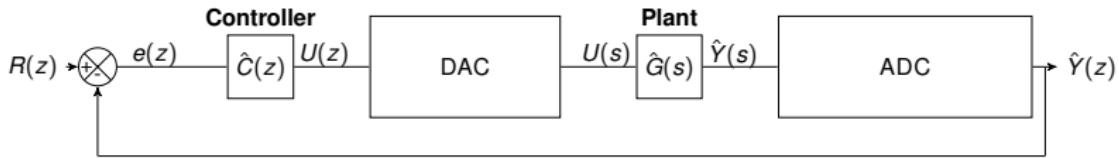
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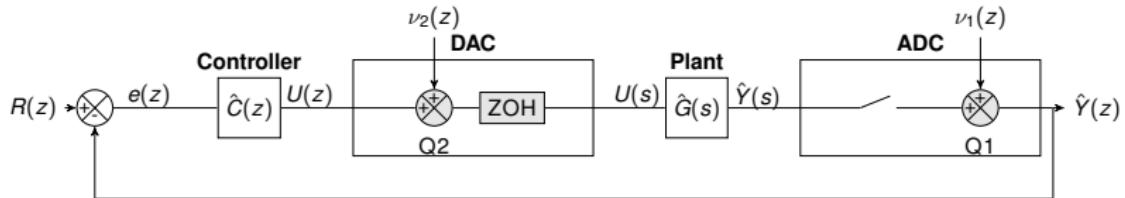


- ▶ use of counterexample guided inductive synthesis (CEGIS) to automate design of sound digital controllers

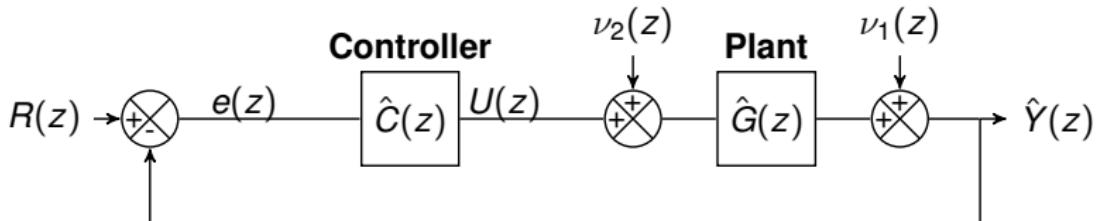
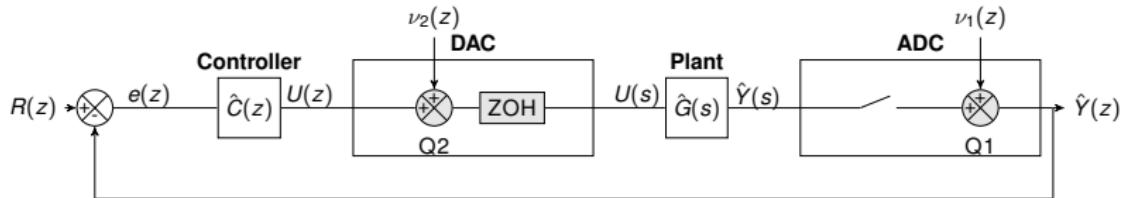
CPS Setup: Continuous Plant and Digital Controller



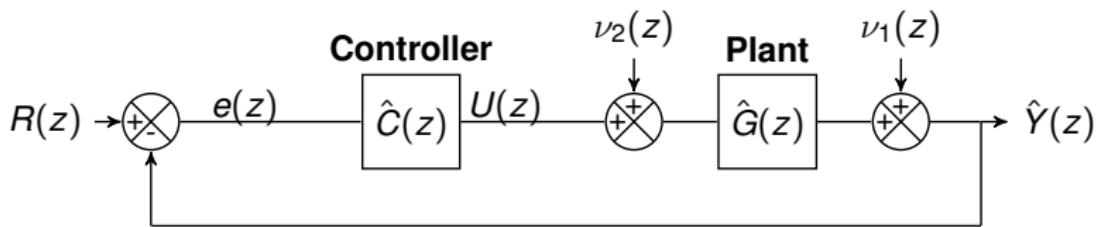
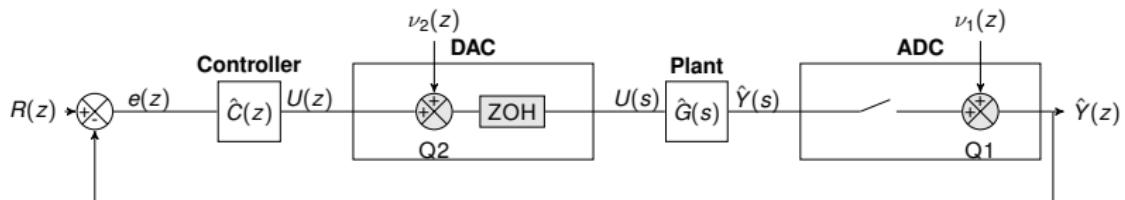
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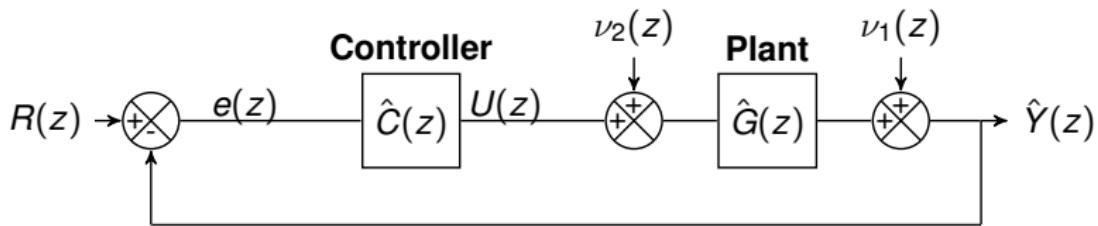
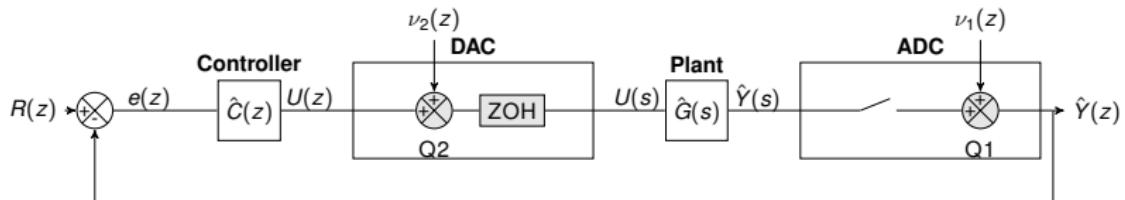


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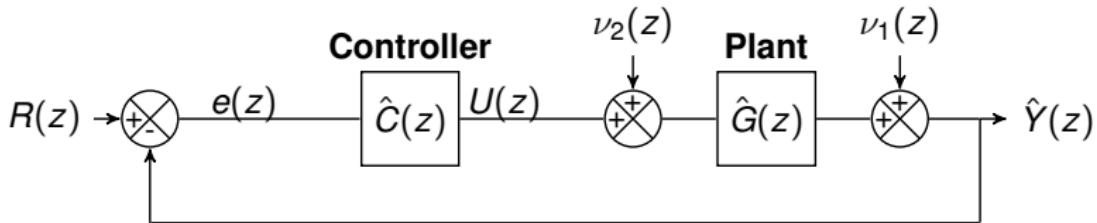
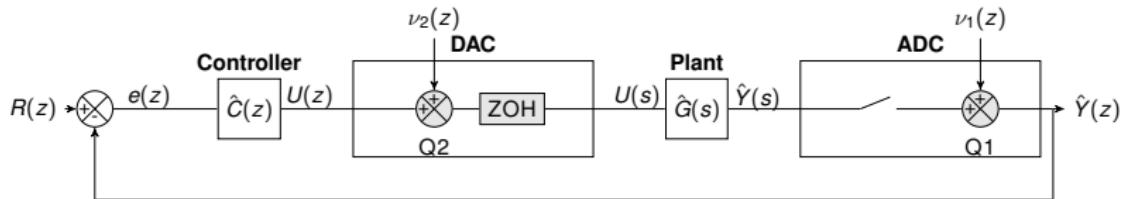
$$\hat{G}(z) = (1 - z^{-1}) \mathcal{Z} \left\{ \mathcal{L}^{-1} \left\{ \frac{\hat{G}(s)}{s} \right\}_{t=kT} \right\}$$

CPS Setup: Continuous Plant and Digital Controller



$$\hat{Y}(z) = \frac{\hat{G}(z)\hat{C}(z)}{1 + \hat{G}(z)\hat{C}(z)} R(z) + \frac{1}{1 + \hat{G}(z)\hat{C}(z)} \nu_1(z) + \frac{\hat{G}(z)}{1 + \hat{G}(z)\hat{C}(z)} \nu_2(z)$$

CPS Setup: Continuous Plant and Digital Controller



Common denominator:

$$\hat{G}_n(z)\hat{C}_n(z) + \hat{G}_d(z)\hat{C}_d(z)$$

Stability Analysis via Jury's Criterion

$$S(z) = \hat{G}_n(z)\hat{C}_n(z) + \hat{G}_d(z)\hat{C}_d(z) = \sum_{i=0}^{N_S} c_i z^{N_S-i}, c_0 \neq 0$$

$$M = \begin{pmatrix} v_1^{(0)} & \dots & v_{N_S}^{(0)} \\ v_1^{(1)} & \dots & v_{N_S}^{(1)} \\ \vdots & \vdots & \vdots \\ v_1^{(N_S-1)} & \dots & v_{N_S}^{(N_S-1)} \end{pmatrix},$$

$$v_j^{(k)} = \begin{cases} c_{j-1}, & k = 0 \\ 0, & k > 0 \wedge \text{if } j > N_S - k \\ v_1^{(k-1)} - v_{N_S-j}^{(k-1)} \cdot \frac{v_1^{(k-1)}}{v_{N_S-k}^{(k-1)}}, & k > 0 \wedge \text{if } j \leq N_S - k \end{cases}$$

$S(z)$ is stable \Leftrightarrow

$$S(1) > 0$$

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Transfer Functions and Uncertainty

$$\hat{C}(z) = \frac{\hat{C}_n(z)}{\hat{C}_d(z)} = \frac{\sum_{i=0}^{M_c} (\beta_i + \Delta\beta_i) z^{-i}}{\sum_{i=0}^{N_c} (\alpha_i + \Delta\alpha_i) z^{-i}},$$

$$\hat{G}(z) = \frac{\hat{G}_n(z)}{\hat{G}_d(z)} = \frac{\sum_{i=0}^{M_G} (b_i + \Delta b_i) z^{-i}}{\sum_{i=0}^{N_G} (a_i + \Delta a_i) z^{-i}}.$$

- ▶ $\vec{\beta}$ and $\vec{\alpha}$ – vectors containing controller coefficients
- ▶ \vec{b} and \vec{a} – plant coefficients

Sources of Uncertainty

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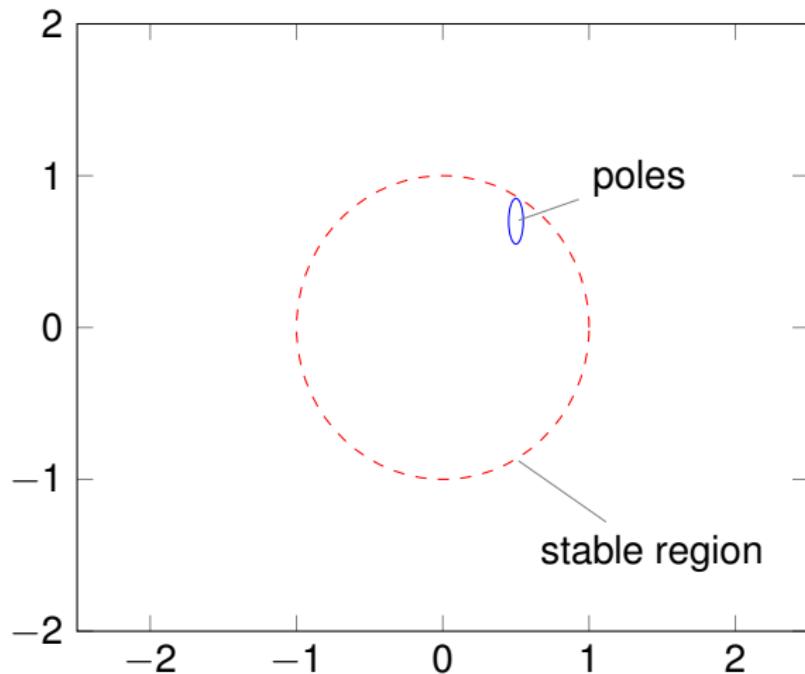
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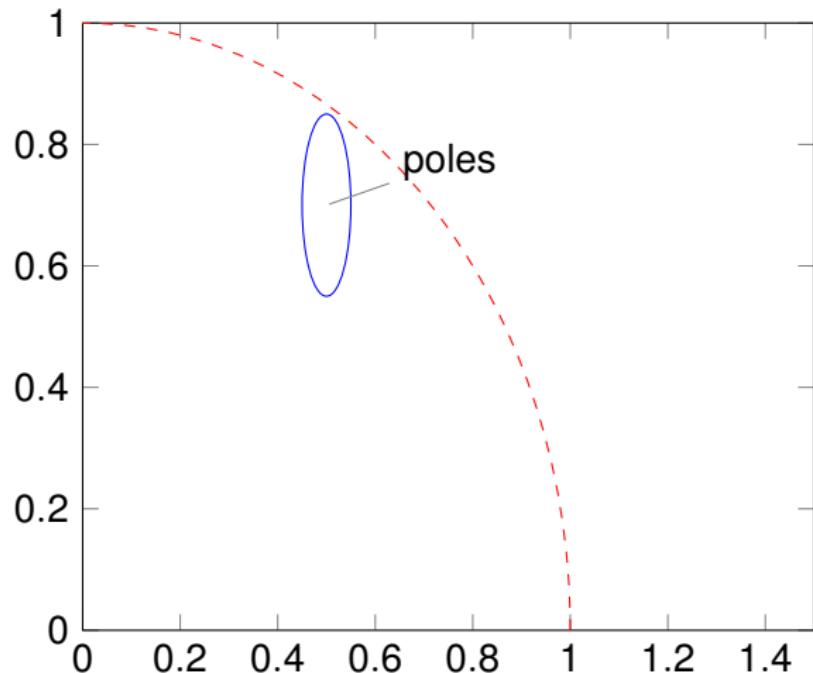
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3. Roundoff errors in verification process (affect $\Delta S(z)$)
4. Discretisation errors (affect $\Delta G(z)$ and $\Delta C(z)$)

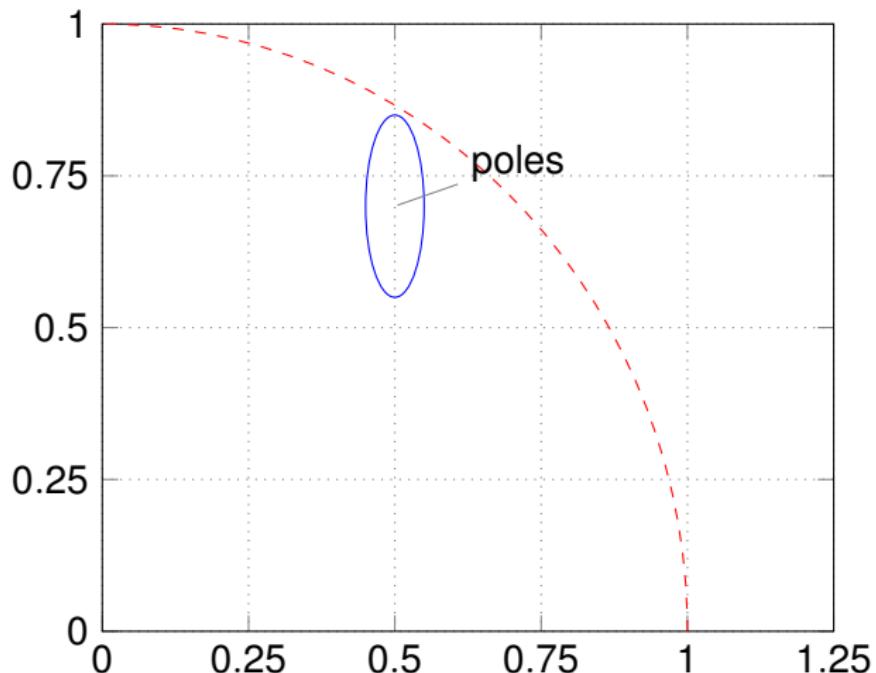
Pole Placement



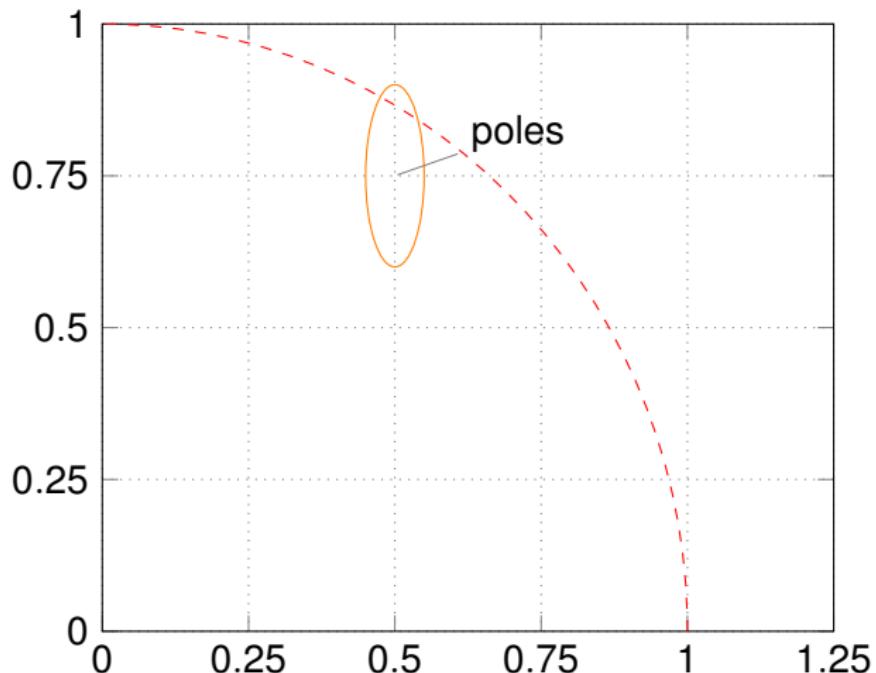
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Pole Placement in a Discretized Domain

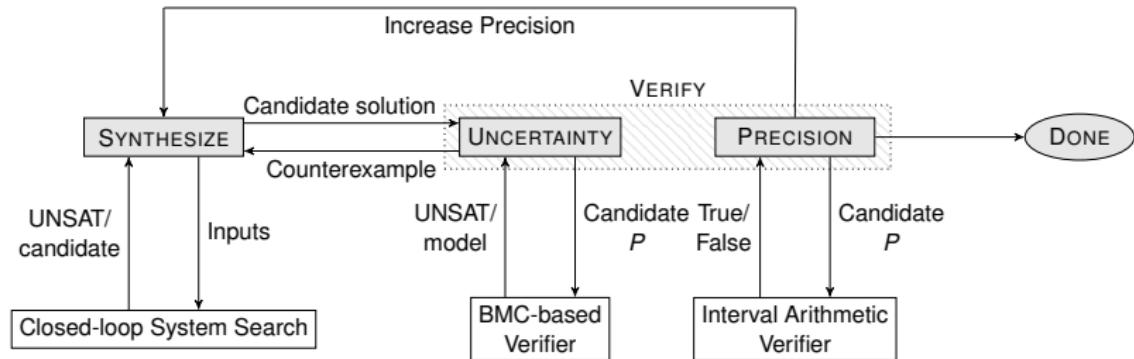


Pole Placement in a Discretized Domain



Objective

We seek to synthesise a controller where $\Delta\vec{\beta} = 0$ and $\Delta\vec{\alpha} = 0$ and which stabilizes all plants within a given range of $\Delta\vec{b}$ and $\Delta\vec{a}$.



Case Study

Take a classical cruise control example from the literature.

$$G(z) = \frac{0.0264}{z - 0.9998}.$$

Using an optimization tool, the following stable high-performance controller was designed:

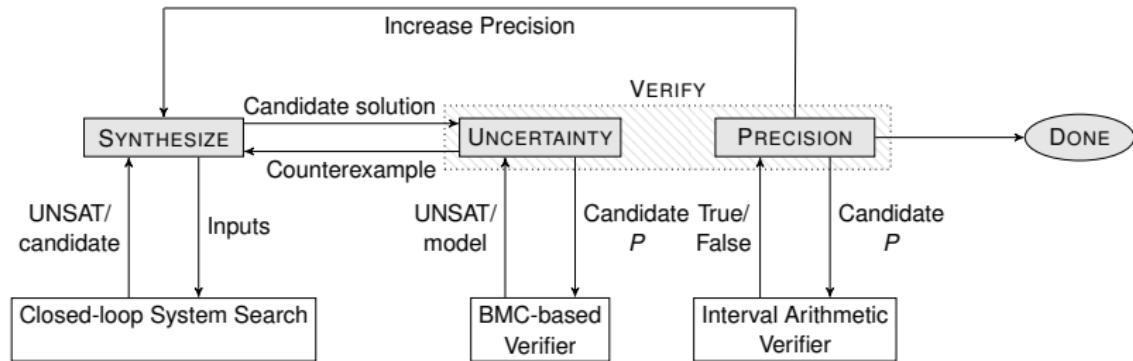
$$C(z) = \frac{2.72z^2 - 4.153z + 1.896}{z^2 - 1.844z + 0.8496}.$$

However, if we implement the controller in $\mathbb{R}\langle 4, 16 \rangle$.

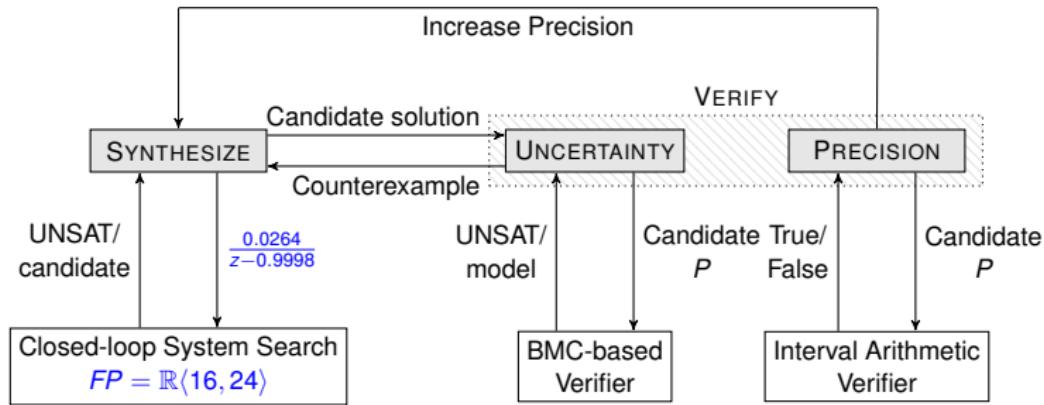
$$\tilde{C}(z) := \frac{2.7199859619140625z^2 - 4.1529998779296875z + 1.89599609375}{z^2 - 1.843994140625z + 0.8495941162109375}.$$

where $\tilde{C}(z)$ is the controller adjusted to the FWL. The resulting system is unstable.

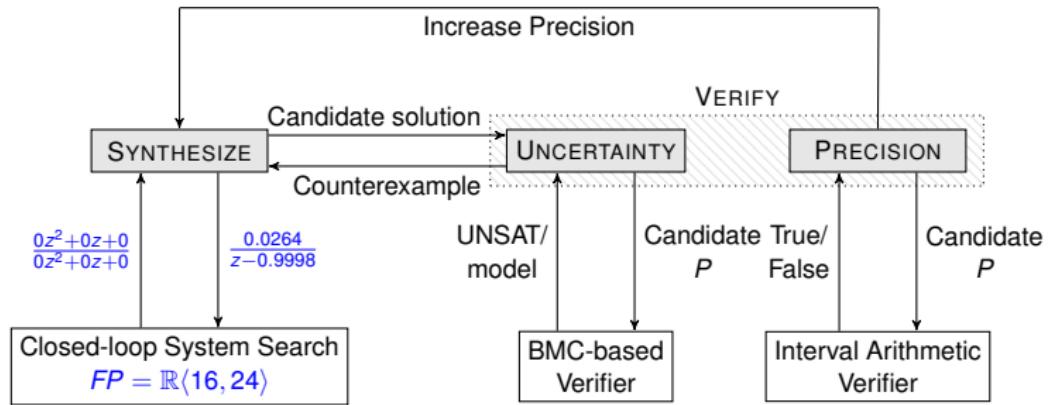
CEGIS Architecture



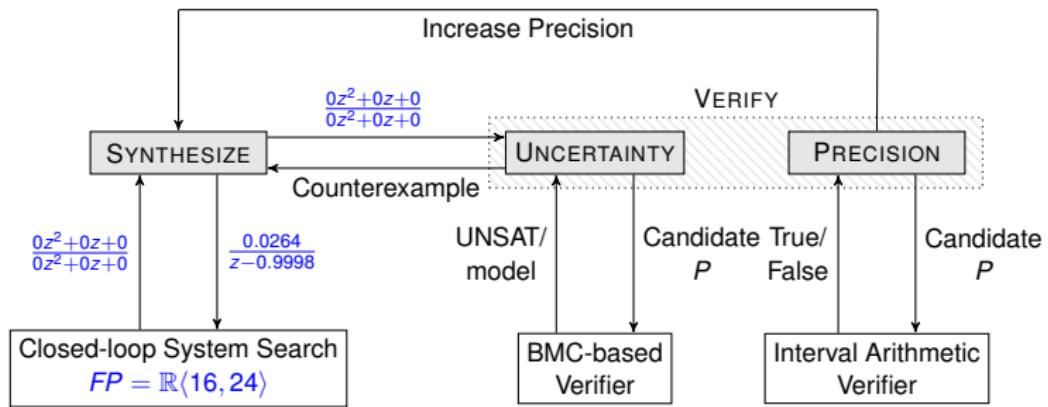
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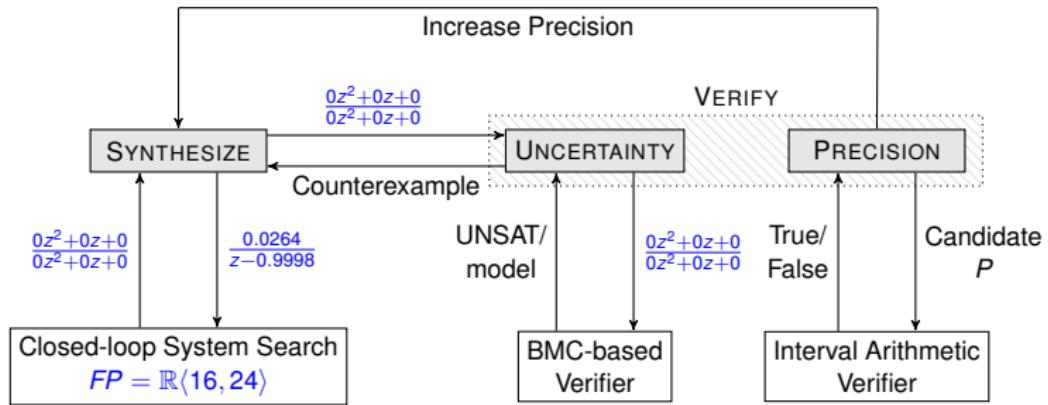
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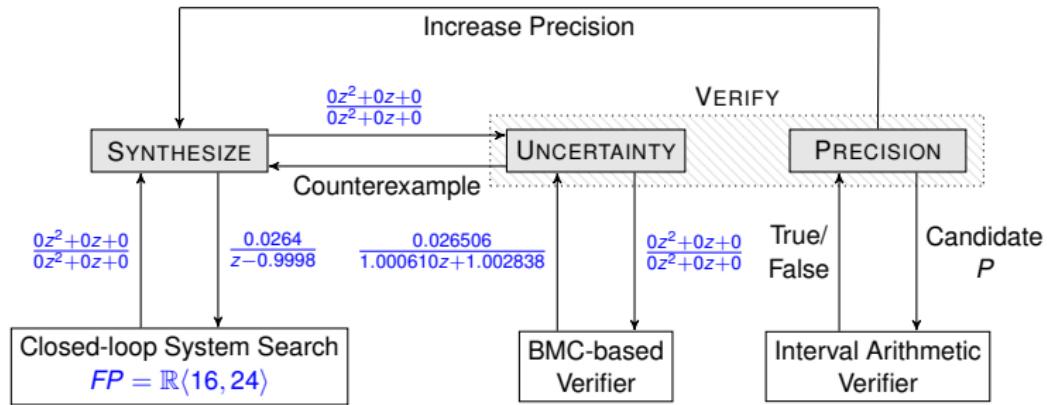
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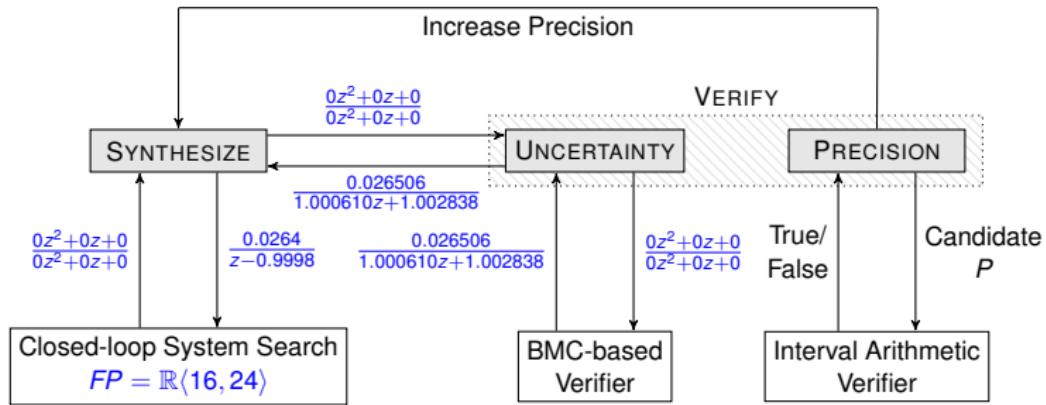
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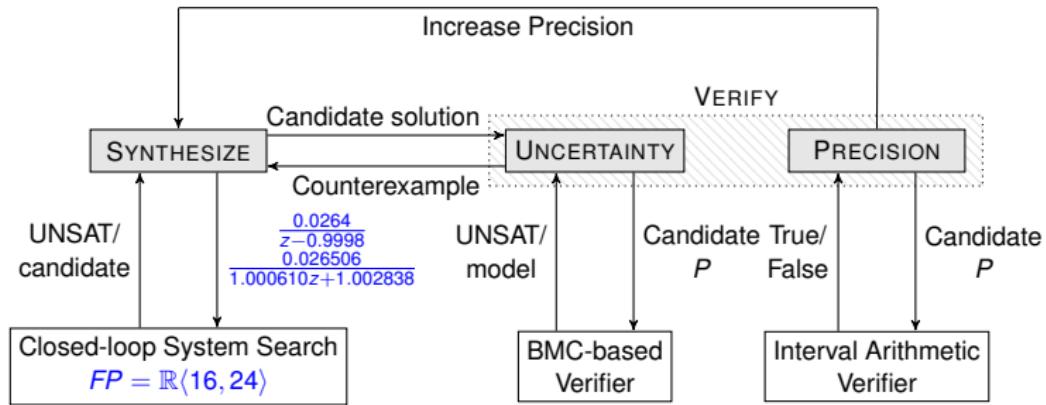
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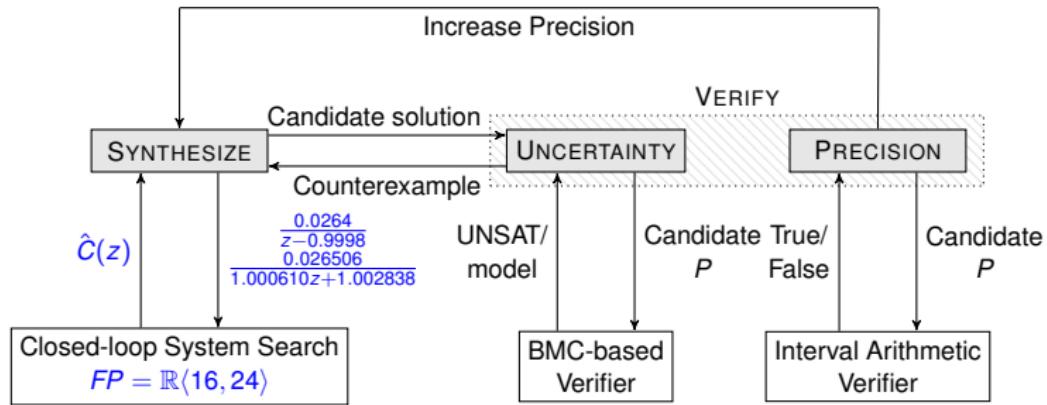
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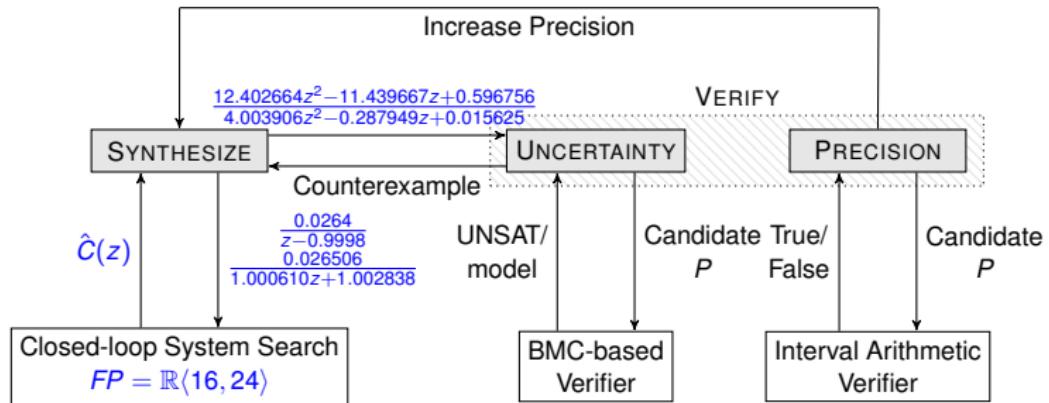
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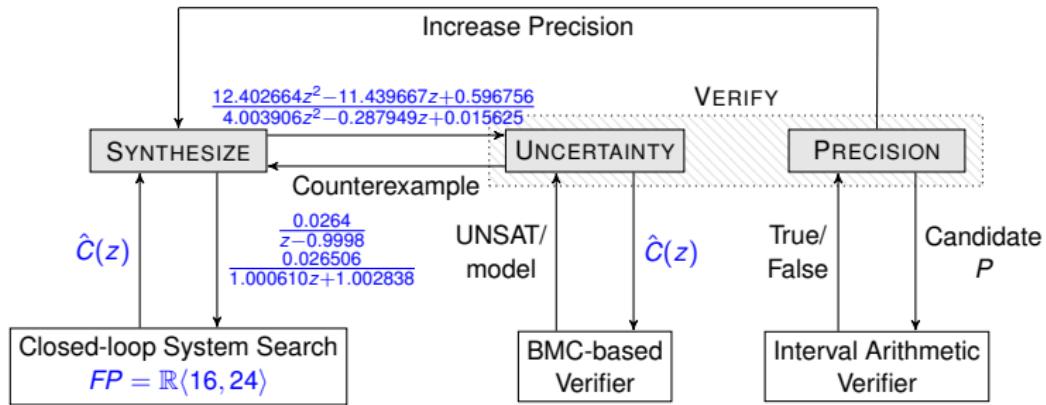
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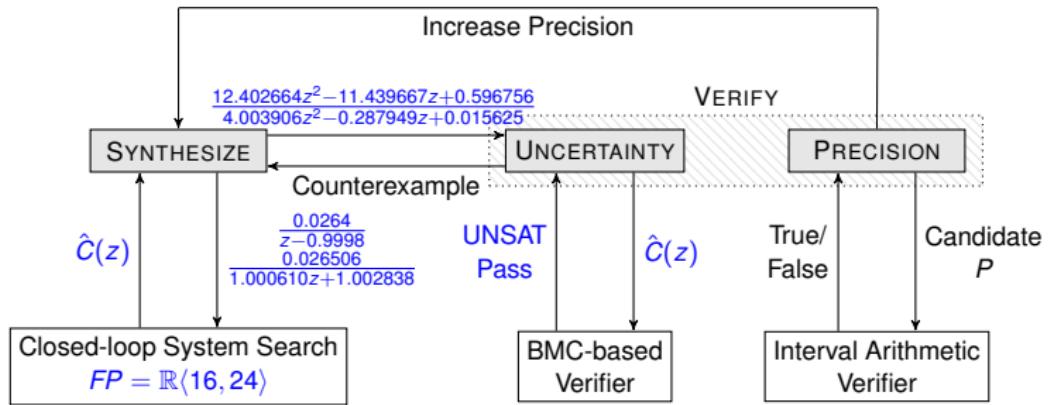
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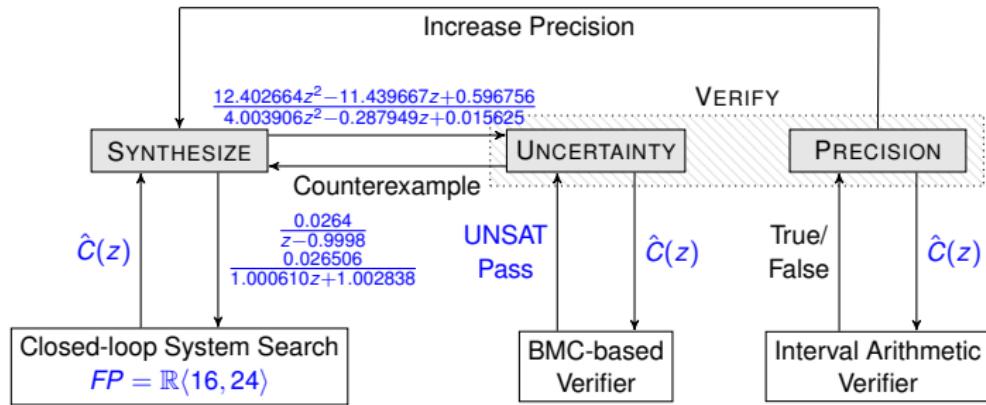
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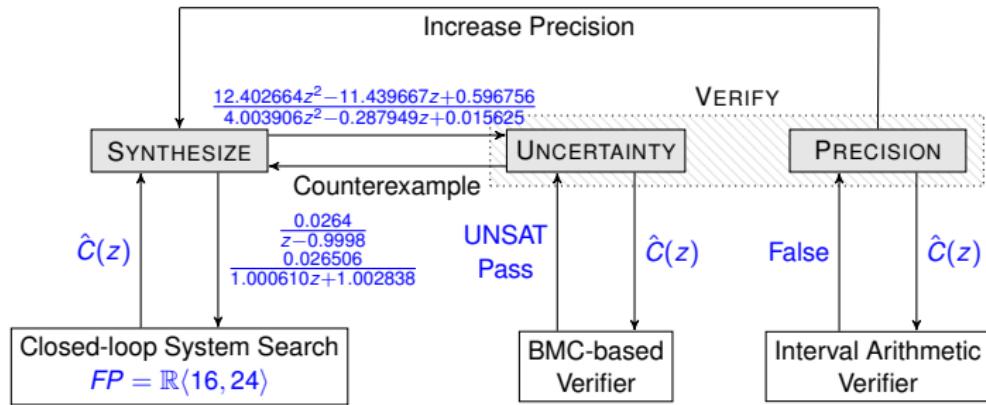
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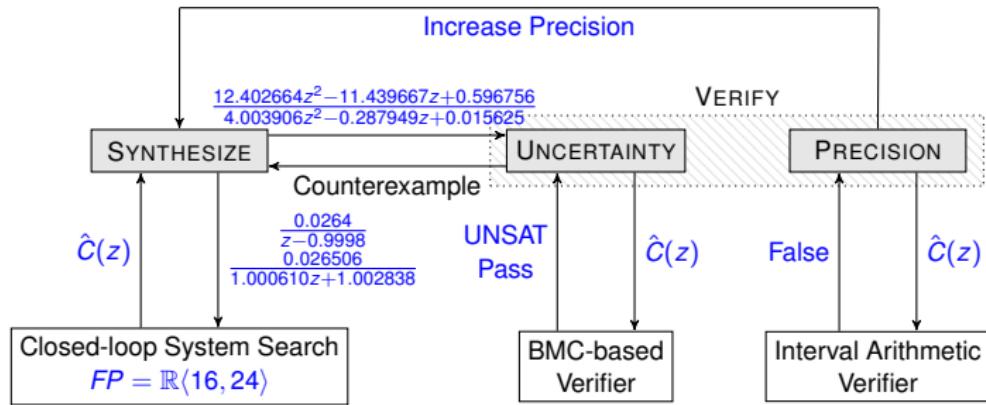
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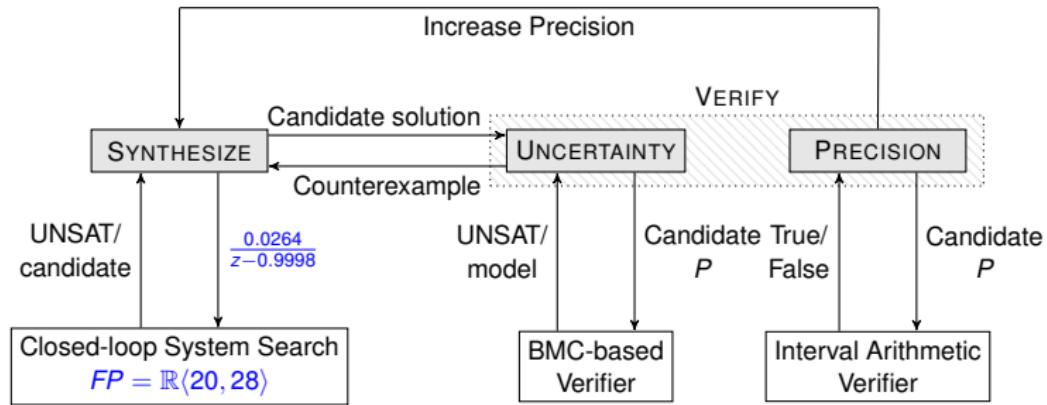
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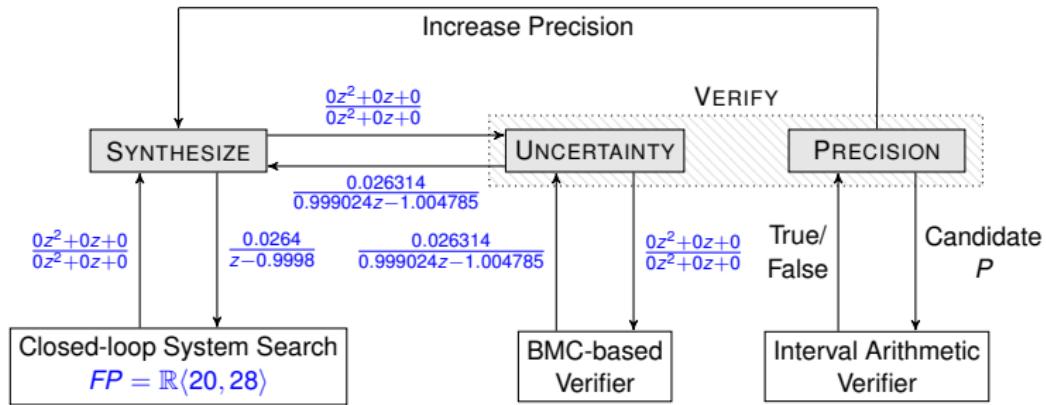
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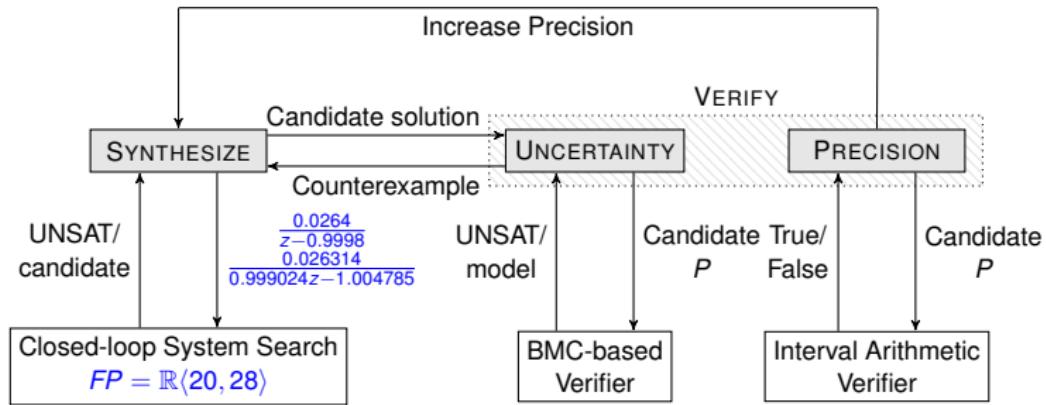
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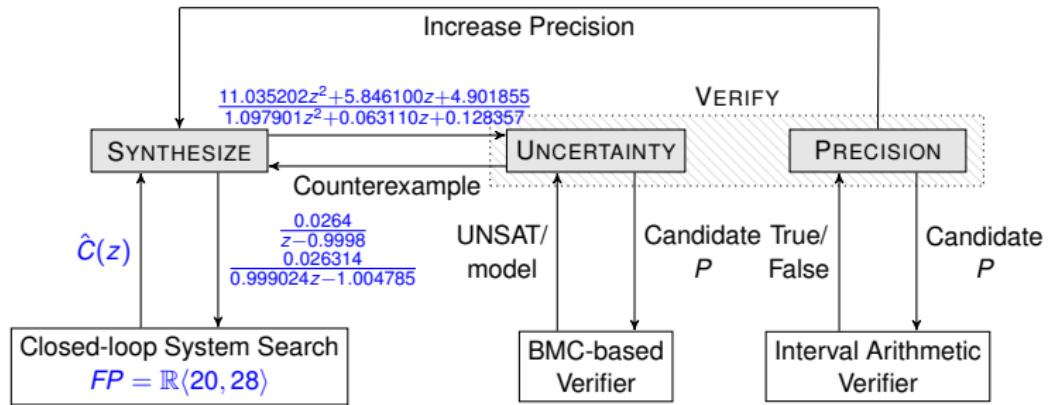
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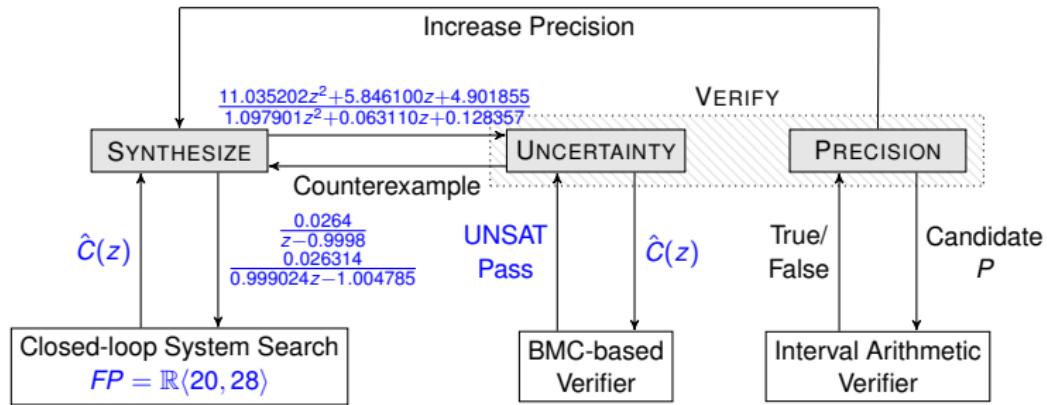
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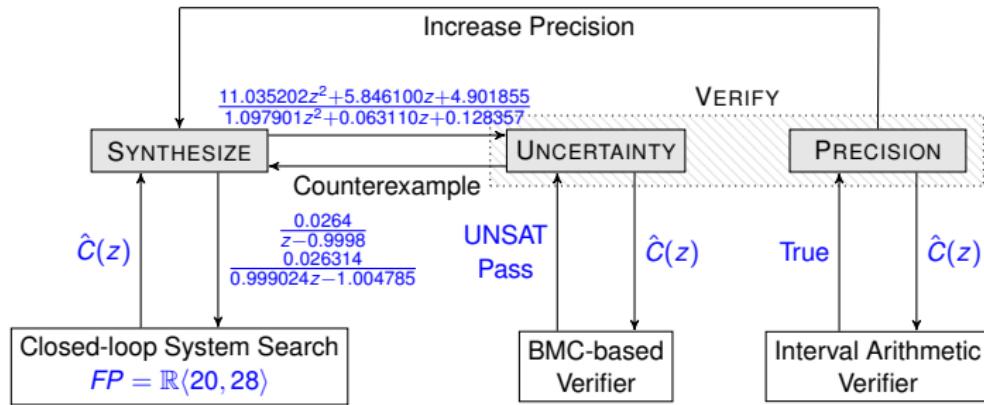
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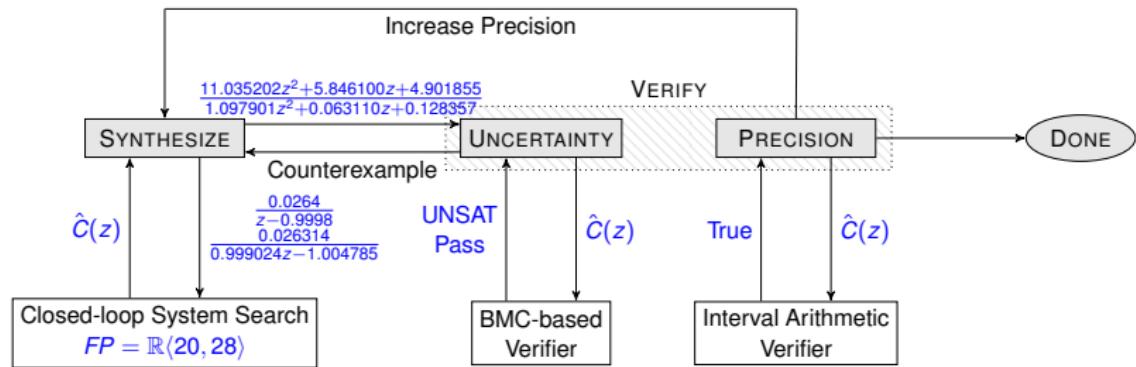
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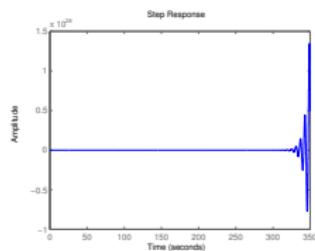
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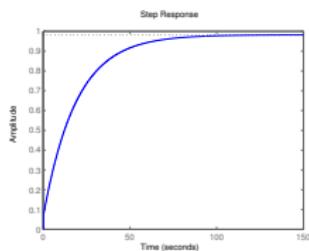
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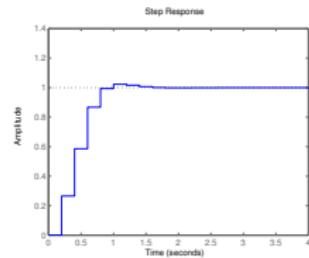
Outcomes: Step Response



(a) Original controller



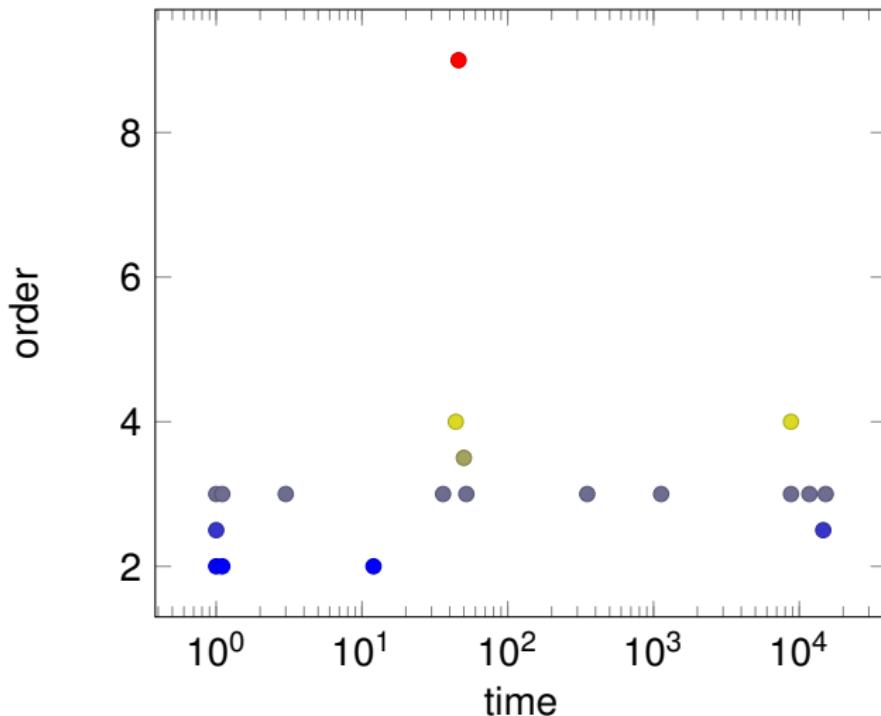
(b) Controller (first step)



(c) Controller (final step)

- ▶ original controller was synthesised ignoring FWL effects
(becomes unstable over time)

Outcomes: Controller Order



Outcomes: Benchmark

#	Plant	Benchmark	I	F	2-stage	1-stage
1	G_{1a}	CruiseControl02	4	16	12s	67s
2	G_{1b}	CruiseControl02 [†]	4	16	14600s	52s
3	G_{2a}	SpgMsDamper	15	16	52s	318s
4	G_{2b}	SpgMsDamper [†]	15	16	X	X
5	G_{3a}	SatelliteB2	3	7	36s	X
6	G_{3b}	SatelliteB2 [†]	3	7	X	4111s
7	G_{3c}	SatelliteC2	3	5	3s	205s
8	G_{3d}	SatelliteC2 [†]	3	5	50s	1315s
9	G_4	Cruise	3	7	1s	1s
10	G_5	DCMotor	3	7	1s	10s
11	G_6	DCServomotor	4	11	46s	X
12	G_7	Doyleetal	4	11	8769s	X
13	G_8	Helicopter	3	7	44s	X
14	G_9	Pendulum	3	7	1s	14826s
15	G_{10}	Suspension	3	7	1s	5s
16	G_{11}	Tapedriver	3	7	1s	1s
17	G_{12a}	a.ST1_IMPL1	16	4	11748s	X
18	G_{12a}	a.ST1_IMPL2	16	8	351s	X
19	G_{12a}	a.ST1_IMPL3	16	12	8772s	X
20	G_{12b}	a.ST2_IMPL1	16	4	1128s	X
21	G_{12b}	a.ST2_IMPL2	16	8	X	X
22	G_{12b}	a.ST2_IMPL3	16	12	15183s	X
23	G_{12c}	a.ST3_IMPL1	16	4	X	X

Table: DSSynth results (X = time-out, [†] = uncertainty)

Conclusions

- ▶ We have presented a method for automatically synthesizing stable and sound controllers, implemented in a tool called DSSynth.
- ▶ DSSynth marks the first use of the CEGIS that handles plants with uncertain models and FWL effects over the digital controller.
- ▶ Our experimental results show that DSSynth is able to synthesize stable controllers for most benchmarks within a reasonable amount of time.
- ▶ Future work will extend this CEGIS-based approach to LTI systems with state space safety specifications.